## REPORT DOCUMENTATION PAGE

Form Approved OMB No. 0704-0188

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing this collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden to Department of Defense, Washington Headquarters Services, Directorate for Information Operations and Reports (0704-0188), 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302. Respondents should be aware that notwithstanding any other provision of law, no person shall be subject to any penalty for failing to comply with a collection of information if it does not display a currently valid OMB control number. PLEASE DO NOT RETURN YOUR FORM TO THE ABOVE ADDRESS.

1. REPORT DATE (DD-MM-YYYY)	2. REPORT TYPE	3. DATES COVERED (From - To)	
06-05-2005			
4. TITLE AND SUBTITLE	5a. CONTRACT NUMBER		
Kinetics of thermal decomposition of decomposition under vacuum and in	5b. GRANT NUMBER		
<b>F</b>	5c. PROGRAM ELEMENT NUMBER		
6. AUTHOR(S)		5d. PROJECT NUMBER	
I.M.K. Ismail (ERC) and T.W. Hawkin	s (AFRL/PRSP)	50260541	
		5e. TASK NUMBER	
		5f. WORK UNIT NUMBER	
7. PERFORMING ORGANIZATION NAME(S	8. PERFORMING ORGANIZATION REPORT NUMBER		
Air Force Research Laboratory (AFMC		KEI OKT NOMBEK	
AFRL/PRSP	,	AFRL-PR-ED-JA-2005-154	
10 E. Saturn Blvd.			
Edwards AFB CA 93524-7680			
9. SPONSORING / MONITORING AGENCY	NAME(S) AND ADDRESS(ES)	10. SPONSOR/MONITOR'S ACRONYM(S)	
Air Force Research Laboratory (AFMC	7)		
AFRL/PRS	,	11. SPONSOR/MONITOR'S	
5 Pollux Drive		NUMBER(S)	
Edwards AFB CA 93524-70448	AFRL-PR-ED-JA-2005-154		
/A DIATRIBUTION / ANY ANY ANY ANY			

#### 12. DISTRIBUTION / AVAILABILITY STATEMENT

Approved for public release; distribution unlimited (PA number AFRL-ERS-PAS-2005-094).

#### 13. SUPPLEMENTARY NOTES

Submitted for publication in Thermochemica Acta, 439 (2005) 32–43

#### 14. ABSTRACT

Recently, interest in aluminium hydride (alane) as a rocket propulsion ingredient has been renewed due to improvements in its manufacturing process and an increase in thermal stability. When alane is added to solid propellant formulations, rocket performance is enhanced and the specific impulse increases. Preliminary work was performed at AFRL on the characterization and evaluation of two alane samples. Decomposition kinetics were determined from gravimetric TGA data and volumetric vacuum thermal stability (VTS) results. Chemical analysis showed the samples had 88.30% (by weight) aluminium and 9.96% hydrogen. The average density, as measured by helium pycnometery, was 1.486g/cc. Scanning electron microscopy showed that the particles were mostly composed of sharp edged crystallographic polyhedra such as simple cubes, cubic octahedrons and hexagonal prisms. Thermogravimetric analysis was utilized to investigate the decomposition kinetics of alane in argon atmosphere and to shed light on the mechanism of alane decomposition. Two kinetic models were successfully developed and used to propose a mechanism for the complete decomposition of alane and to predict its shelf-life during storage. Alane decomposes in two steps. The slowest (rate-determining) step is solely controlled by solid state nucleation of aluminium crystals; the fastest step is due to growth of the crystals. Thus, during decomposition, hydrogen gas is liberated and the initial polyhedral AlH3 crystals yield a final mix of amorphous aluminium and aluminium crystals. After establishing the kinetic model, prediction calculations indicated that alane can be stored in inert atmosphere at temperatures below 10°C for long periods of time (e.g., 15 years) without significant decomposition. After 15 years of storage, the kinetic model predicts ~ 0.1% decomposition, but storage at higher temperatures (e.g. 30°C) is not recommended.

#### 15. SUBJECT TERMS

16. SECURITY CLASSIFICATION OF:			17. LIMITATION OF ABSTRACT	18. NUMBER OF PAGES	19a. NAME OF RESPONSIBLE PERSON	
					Dr. Tommy W. Hawkins	
a. REPORT	b. ABSTRACT	c. THIS PAGE			19b. TELEPHONE NUMBER	
			Α	18	(include area code)	
Unclassified	Unclassified	Unclassified		10	N/A	

# Kinetics of thermal decomposition of aluminum hydride: I- non-isothermal decomposition under vacuum and in inert atmosphere (argon)

## I. M. K. Ismail and T. Hawkins

ERC, Inc\* at Air Force Research Laboratory/PRSP, Edwards Air Force Base, CA 93523-76891

#### Abstract

Recently, interest in alane (aluminum hydride) has been renewed as a rocket propulsion ingredient due to improvements in its manufacturing process and an increase in thermal stability. When alane is added to solid propellant formulations, rocket performance is enhanced and the specific impulse increases. Preliminary work was performed at AFRL on the characterization and evaluation of two alane samples. Decomposition kinetics have been determined from gravimetric TGA data and volumetric vacuum thermal stability (VTS) results. Chemical analysis showed the samples had 88.30% (by weight) aluminum and 9.96% hydrogen. The average density, as measured by helium pycnometery was 1.486 g/cc. Scanning electron microscopy showed that the particles were mostly composed of sharp edged crystallographic polyhedra such as simple cubes, cubic octahedron and hexagonal prisms.

Thermogravimetric analysis was utilized to investigate the decomposition kinetics of alane in argon atmosphere and to shed light on the mechanism of alane decomposition. Two kinetic models have been successfully developed and used to propose a mechanism for the complete decomposition of alane and to predict its shelf-life during storage. Alane decomposes in two steps; the slowest is solely controlled by solid state nucleation of aluminum crystals; the fastest is due to growth of the crystals. Thus, during decomposition, hydrogen gas is liberated and the initial polyhedra AlH<sub>3</sub> crystals yield a final mix of amorphous aluminum and aluminum crystals. The nucleation of aluminum crystals is the rate determining step. After establishing the kinetic model, prediction calculations indicated that alane can be stored in inert atmosphere at temperatures below 10°C for long periods of time (e.g. 15 years) without significant decomposition. After 15 years storage, the kinetic model predicts ~ 0.1% decomposition but storage at higher temperatures (e.g. 30°C) is not recommended.

*Keywords*: Alane; Aluminum hydride; Shelf-life prediction; Decomposition; Model free kinetics; MFK; Thermokinetics; Thermal stability; VTS

#### 1. Introduction

Alane, AlH<sub>3</sub>, is a solid rocket fuel containing (theoretically) 10.08 % by weight hydrogen. It was evaluated in the early 1960's for military applications and was formulated in propellants. This process enhanced the development of cracks in the solid propellant. Rocket motor firing was successfully completed up to 300-Lb motors. However, the low thermal stability of the alane used at that time resulted in the generation of excessive amounts of hydrogen gas in the solid propellant during storage. By the early 1970's, most of the development work in the USA came to an end.

Recently, the interest in alane has been renewed presumably after the development of new methods of preparations and after the discovery of new stabilizers that can slow the rate of alane decomposition and thus increase its shelf-life. With the concern on alane thermal stability, extensive physical and chemical characterization of newly produced alane is needed, especially in the area of thermal stability and hydrogen generation during long term storage.

There have been a number of publications in the open literature on preparation, characterization, decomposition kinetics and vacuum thermal stability (VTS) of alane. Depending on the method and conditions of preparation (mainly reaction time and temperatures), alane can be produced in at least six different crystal forms [1]. The most stable form is  $\alpha$ -alane which has been used in the current

<sup>&</sup>lt;sup>1</sup> Approved for public release; distribution unlimited.

investigation. Other forms are  $\beta$ -alane and  $\gamma$ -alane which are meta-stable phases. Brower *et.al.* [1] reported that in the final steps of aluminum hydride synthesis, the  $\beta$  and  $\gamma$  forms were first formed; upon further heating, they were converted to  $\alpha$ -alane. Polarizing-light microscopy showed the crystals of the  $\alpha$ -phase were mostly hexagonal and cubical. The  $\gamma$ -phase appeared as bundles of fused needles in random orientation. Another phase,  $\gamma$ '-alane appeared in the examination as small multiple needles growing from one center and forming fuzzy balls [1]. From VTS experiments [1], determined by the Taliani method [2], the  $\alpha$ -phase was the most stable form; typically its crystals (50– 150  $\mu$ ) decomposed between 0.5 and 2.0% after 6 days at 60°C depending on particle size.

The heat of decomposition of aluminum hydride samples have been measured experimentally in nitrogen atmosphere using a modified bomb calorimeter accommodating a small suspended heating oven containing the sample [3]. At 298 K, the calculated average enthalpy of formation was -11.4  $\pm$  0.8 kJ/mol, absolute entropy was 30.0  $\pm$  0.4 kJ/mol °C and Gibbs energy of formation was 45.4  $\pm$  1.0 kJ/mol. These values indicate that alane is an unstable compound with respect to its forming elements. Thus, thermodynamically, alane should naturally decompose to yield aluminum metal and hydrogen gas.

The VTS of un-stabilized and stabilized aluminum hydride samples have been measured at 60°C [4]. The objectives were (a) to develop new methods for alane synthesis and (b) to develop new methods to stabilize the product. The time taken to reach 1% level of decomposition was ~ 9.5 days for the original sample and ~ 12.7 days for the stabilized one. It is interesting to add that for many investigators, the VTS tests have been performed at 60°C until reaching a 1% level of decomposition by weight; these test conditions are somewhat arbitrary. In most cases, most of the work was performed using the original "old" Taliani method [2] or a modified one. The kinetics and mechanism of the thermal decomposition of several solvated aluminum hydride compounds have been reported by Zakharov and Tskhai [5]. The volume of decomposition products was traced as a function of time at several isothermal temperatures between 50 and 100°C. For dry alane samples (that is, after the removal of the solvents), an s-shaped type plot was obtained during the liberation of hydrogen. The kinetics of hydrogen liberation was described by a first order autocatalytic equation. The activation energy of this step was 72.2 ± 2.5 kJ/mol.

The thermal decomposition of AlH<sub>3</sub> and AlD<sub>3</sub> was investigated using NMR [6]. Samples of alane were decomposed isothermally (at 86-127°C) and typical s-shaped plots, correlating percent decomposition (or amount of aluminum produced) versus reaction time, t, were obtained. Each s-curve was divided into three regions; the first of which corresponded to an induction period,  $\tau_{ind}$ , where the rate constant  $k_1$  =  $1/\tau_{ind}$ . The value of  $\tau_{ind}$  was defined as the time taken to achieve 5% decomposition (or when the fraction of original alane converted to aluminum,  $\alpha$ , is 0.05). In the second region, an "acceleration period" [6] was noted between  $\alpha = 0.1$  and 0.6; the equation governing the kinetics in this region was  $\alpha^{1/3} = k_2 t$ . The third region,  $\alpha = 0.6 - 0.9$ , was defined as the deceleration region, the kinetic rate was controlled by a third equation:  $\ln(1-\alpha)^{-1} = k_3 t$ . The activation energies for the three regions were 97, 108 and 112 kJ/mol, respectively, within an error of 20% [6]. A comparison between decomposition kinetics of AIH<sub>3</sub> and AID<sub>3</sub> showed that the deuteride was decomposing at a slower rate than the hydride. For example, at 107°C, the induction period for the deuteride was 5 times longer than that of the hydride, and the rate constant for the deuteride (presumably in the second region) was half of that for the hydride [6]. The differences were attributed to the presence of different impurities and crystal defects in each sample and also to differences in particle size. The effect of particle size of the hydride at 107°C was studied between 50 and 150 µ, decomposition kinetics were highly dependent on particle size [6]. For example, to achieve the complete liberation of hydrogen, it took  $\sim$  350 min for the sizes > 150  $\mu$  but only  $\sim$  90 min for the 50  $\mu$ particles. The implication of this is that the thermal stability of alane can be enhanced, to a point, simply by increasing the starting particle size of alane.

Over the past decade, significant improvement was made to produce kinetic modeling software that help investigators to better understand the kinetics and mechanism of chemical reactions [7-19]. These programs proved to be useful in many applications. Typical commercial software are available such as "Model Free Kinetics" by METTLER or "Thermokinetics" by NETZSCH (which was used for the work cited in this article) and AKTS-TA-Software. Basically these programs perform sophisticated computation in a logical order. For example, a set of experiments can be performed on the sample at different heating rates (*non-isothermal method*). The iso-conversion (i.e., at constant values of  $\alpha$ ) method can normally be used to start the analysis; it yields information on how the activation energy of the entire process (or its individual steps), and the pre-exponential factor(s) could be changing as the reaction progresses, that is

from  $\alpha=0$  to 1.00. With these results taken into consideration, the programs determine numerically the best kinetic model that fits the data mathematically. The Thermokinetics program has several mathematical models that prescribe chemical reactions and solid state transformations. The models include  $n^{th}$  order reactions, Avrami-Erofeev solid state transformation [20] and autocatalytic reactions. A single one-step model or a combination of models (more than one step reaction) can be chosen for performing the analysis. These software programs have been used in the recent literature to investigate the thermal decomposition of many energetic materials including HMX [10-12], ammonium perchlorate [13], ammonium nitrate [14] and ammonium dinitramide [15-16]. The programs were also used to study dimerization of cyclopentadiene [17], to predict the thermal response of hazardous materials during storage [18] and the thermal reactions of rocket motor propellants [19]. This type of analysis yields a wealth of information on decomposition kinetics and mechanisms of these ingredients and propellants. To our knowledge, a similar investigation or study of alane decomposition kinetics has not been reported in the open literature.

The first objective of the present work was to develop kinetic equations (or a model) that can accurately describe the decomposition process of alane in an inert atmosphere. The second objective was to use the model and propose a mechanism for alane decomposition. The third objective was to use the model and predict the effective storage life of alane when stored at ambient conditions.

## 2. Experimental

#### 2.1 Materials

The alane samples used in the present investigation had the following typical elemental analysis: average weight percentage of aluminum was 88.26%, of hydrogen 9.96% and of carbon 0.44%. The balance, obtained by difference, 1.54% was assigned to oxygen.

#### 2.2 Vacuum Thermal Stability (VTS)

A new twin-chamber VTS apparatus (Figure 1) was constructed to investigate the thermal stability of alane. Each chamber has the following major components: a Pyrex tube containing the sample which is contained inside a stainless steel tube. The tube was connected to a pressure transducer and a fine metering valve. Two type-K thermocouples were attached on the outer skin of the reactor. The entire system (A or B) was placed inside an oven regulated at a chosen temperature (normally  $60^{\circ}$ C). The samples were evacuated at room temperature for 24 hours using a combination of a roughing pump and an oil diffusion pump attached to a cryogenic liquid nitrogen trap. This system was able to pump the samples down to better than  $10^{-7}$  torr. The metering valve was closed and the entire system (reactor and all other accessories) was suspended inside the oven. The electrical wirings for the components passed through a hole at the top of the oven and were connected to two data acquisition modules and a power supply. The modules were connected to feed the serial port of a laptop computer. The experimental data were saved every two minutes; including one pressure reading and two temperature readings as a function of time. A run was typically continued until the sample reached the desired level of decomposition (~1% weight loss). These analyses often took 5-12 days.

#### 2.3 Thermal analysis (DSC and TGA experiments)

A Mettler TGA unit (model SDTA-851e) was used with Mettler proprietary software STAR<sup>e</sup> version 8. Ultra-high-purity argon (UHP certified purity: 99.9999% Ar) was passed through the samples at a flow rate of 60 cc/min. For the present work, special sample containers were fabricated of quartz; the assumption was made that the regular alumina sample containers used with TGA may affect the results. Because of the differences in thermal properties of quartz and those of the other materials, the calibration coefficients of TGA were determined using the quartz containers, as per the Mettler guidelines.

Before starting a test, a sample was flushed with the selected gas for 20 minutes to eliminate residual air present in the system. The sample was then heated at a projected constant heating rate, between 1 and 20 C/min from room temperature to 350°C. After terminating a run, the density of residue was determined using a Micromeritics helium micro-Accu-Pycnometer (cell volume: 0.25 cc and expansion volume: 0.73 cc). The surface morphology of residues was examined using SEM techniques.

The surface area and porosity development of one alane sample was studied in the following manner, using an automated surface area apparatus, Digisorb 2600, manufactured by Micromeritics Instrument Corporation. Successive runs were performed on one sample (starting weight = 0.85 g) while alternating between a pre-evacuation step at 60°C for a given period of time, and actual measurements of surface area using krypton adsorption at -196°C. The sample was always kept under vacuum, in He or in Kr without ever being exposed to ambient air for the entire set of tests. This practice insured that the generated aluminum did not oxidize during the measurements. Details of this test are given next.

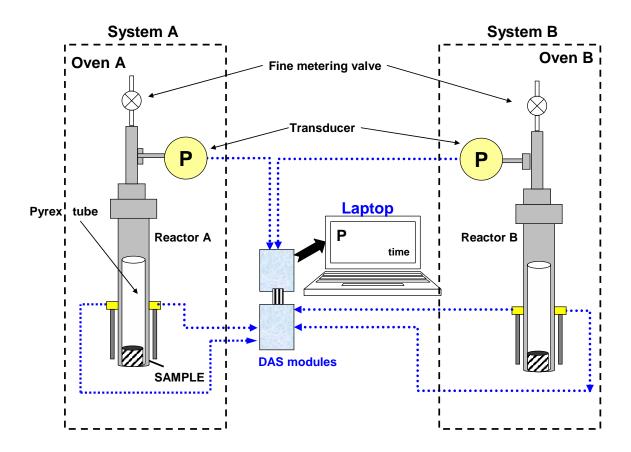


Fig.1. Vacuum thermal stability dual system.

## 2.4 Calculations

Specific surface area, SA  $(m^2/g)$ , was calculated from Kr adsorption isotherms obtained at liquid nitrogen temperature using the BET equation. The cross-sectional area of a Kr atom adsorbed at the surface was taken as  $0.210 \text{ nm}^2/\text{Kr}$  atom [21].

For the TGA results, the data of alane decomposition were evaluated using NETZSCH-Thermokinetics software supplied by NETZSCH-Gerätebau GmbH in Germany.

#### 3. Results and Discussion

#### 3.1 Characterization of as-received alane

Selected two alane samples were sent to Galbraith Labs for elemental analysis (AI, H, and C assays). The results are summarized in Table 1 for the dried samples. Pure alane has a stoichiometric composition corresponding to 89.92% Al and 10.08% H. It is noted in the table that the two samples are short in Al and H because they contain other species such as C and possibly oxygen. The stoichiometric Al/H ratio is 8.921 for pure alane; the average experimental ratio was 8.86. The samples have excess (organic) H that could not be accounted for. The total % of Al plus H plus C did not add up to 100%; the balance can be assigned to other species; mostly oxygen. It was in the range of 1.42% to 1.21%.

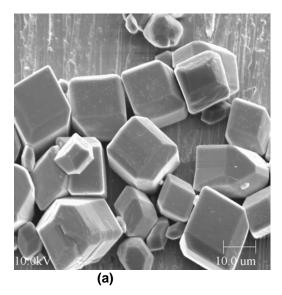
Table 1
Chemical analysis and properties of alane

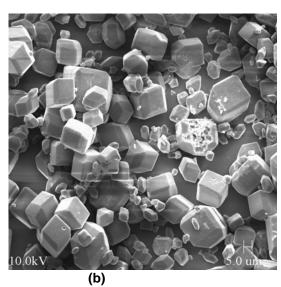
Chemical Analysis	Sample 1	Sample 2
Al, %	88.26	88.34
H, % C, %	9.95 0.37	9.97 0.48
Total	98.58	98.79
AI/H ratio	8.87	8.86
Density, g/cc	1.486	-
Surface Area, m <sup>2</sup> /g	0.215	-

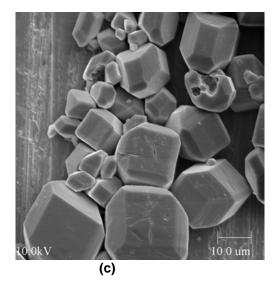
## 3.2 Surface area, morphology and particle size

The surface area of as-received alane, determined by krypton adsorption, was 0.215 m $^2$ /g. This value is small and suggests that alane particles are non-porous or, at least, have insignificant level of porosity. Figure 2 shows the SEM photos of the two alane samples at two magnifications. These particles are not spherical as it has been noted in previous published work on alane. Instead, the particles are mostly cubes with fewer pentagonal and hexagonal crystal shapes. Occasionally, cracks and pits are noted on the surface but these are not deep. The SEM size of the particles ranges between 3 and 20  $\mu\mu$ .

Assuming that all particles are cubical, the average size of particles, and the average equivalent length of its side, L (in microns), can be estimated using the relation: L = 6 / (surface area x density). The average calculated value of L is 18.8  $\mu$  which is in reasonable agreement with SEM observations shown in Figure 2. It is also noted in the SEM photos that most of the particles have smooth surfaces suggesting insignificant porosity.







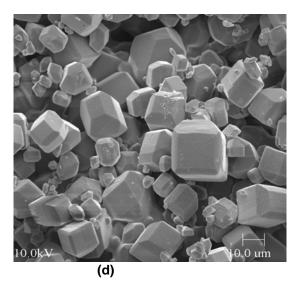


Fig. 2. SEM photos of alane at different magnifications.

#### 3.3 Vacuum thermal stability (VTS)

The effect of pre-evacuation time on decomposition kinetics of alane was first studied. Several samples were first pre-evacuated at room temperature for varying periods of time: 1, 2, 10 and 16 days. There was no effect of pre-evacuation time at room temperature on subsequent decomposition kinetics at  $60^{\circ}$ C. Therefore, a reasonable 24 h pre-evacuation time was chosen for the rest of the work reported here. Following evacuation, the sample was moved to the oven then heated to and held at  $60^{\circ}$ C for several days. The increase in reactor pressure due to hydrogen evolution, and the corresponding percent of sample weight loss were monitored with time. When the sample reached 1% weight loss, the experiment was normally terminated and gas analysis were made using a mass-spectrometer. The main product gas was  $H_2$  containing small trace amounts of water.

Selected results on VTS of alane are shown in Figure 3. For the 4 runs performed, the results were reproducible and the time taken to reach the 1% level of weight loss was in the range of 6.8 to 7.6 days. An average time of  $7.2 \pm 0.4$  days can be confidently assigned to this sample.

#### 3.4 Development of porosity during alane decomposition

Theoretically, when alane completely decomposes to yield aluminum, the density increases from 1.486 to 2.71 g/cc. These values were confirmed in the present investigation on the starting alane and on the final products, using helium pycnometery. From the published crystallography data, the crystal size and structure should also be changing from hexagonal alane (lattice parameters: a=0.54 nm and c=1.14 nm) [22] to face centered cube (FCC) aluminum (a=0.404 nm). Between these two extremes, one would expect a combination of the two structures with the alane crystals breaking down and the growth of aluminum crystals. To shed more light on this effect, the following series of surface area measurements and porosity evaluations were performed on alane. A fresh alane sample ( $\sim 0.85$  g) was evacuated at room temperature for 24 h; it lost 2.28%. The first Kr-surface area measurement at liquid nitrogen temperature was measured: 0.215 m $^2$ /g. The sample was evacuated again at room temperature for additional 24 h, there was no additional weight loss, and the surface area remained the same. The sample was again evacuated for additional 72 h. Again, there was no weight loss or change in surface area. This means that the 2.28% loss after the first evacuation was due to evolution of moisture, trapped solvent (if any) and other volatiles.

The sample was then evacuated stepwise at 60°C for successive periods of time. After each evacuation, the sample was kept under a blanket of ultra high purity He to avoid possible contamination with air and to eliminate the possibility of oxidizing the fresh nascent aluminum active sites developed during hydrogen liberation. After each He fill up, the sample was weighed and a Kr surface area measurement was executed. These two consecutive steps were repeated several times until the final

sample weight became constant; corresponding to the complete decomposition of alane. That is, the remaining weight was  $\sim 90\%$  of the starting weight. The results are displayed in Figure 4. The plot has three main regions. In the first region, between 0 and 1.9% weight loss, the surface area is essentially the same. In this region there is probably loss of hydrogen atoms attached to the external surface of alane cubes plus those attached to the internal walls of open cracks, pits and crevices. The alane particles in this region are, most likely, intact; their sizes are essentially the same as the starting alane. In the second region, between  $\sim 2$  and 7.5% weight loss, the particles develop porosity as a result of the excessive hydrogen gas pressure exerted on pore walls. The walls begin cracking, the pores begin opening and the original crystal structure is opened. As more pores continue to open, the surface area keeps increasing until a maximum is reached at  $\sim 7.5\%$  weight loss. In the third region, the surface area drops abruptly to  $\sim 0.35$  m $^2$ /g and remains constant at this value. This suggests that the final cubical structure of aluminum produced may only dominate after 75% of the original hydrogen evolves.

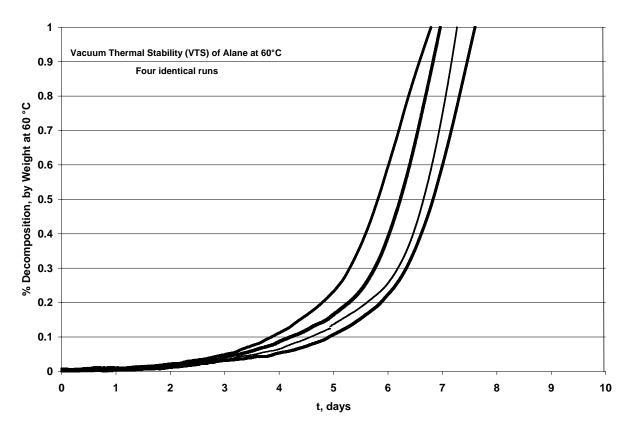


Fig. 3. Vacuum thermal stability of an alane sample at 60°C.

The ratio between the final specific surface area of Al and starting surface area of alane was approximately 1.63. This increase is attributed to two main reasons: a decrease in particle size after decomposition and an increase in surface roughness during decomposition. To verify this, the surfaces of several residues of partially and fully decomposed alane samples were examined using SEM. Typical representative photos are shown in Figure 5. Plate (a) shows that the particles have developed pores and cracks after 3.1% weight loss, and that the external surface acquired slight roughness. Plate (b) shows the surface after the complete decomposition of alane, the roughness has increased drastically and there are small debris of (probably) aluminum metal covering the surface. The average particle size, for the larger particles in the completely decomposed sample, is the range of 12-16  $\mu$ , compared to 18-20  $\mu$  for the starting alane. The most interesting observation is that the original cubical and other geometrical shapes of the starting alane particles are still preserved in the final product; cubical aluminum. The experimental density of final residue after complete decomposition (plate b) was 2.69 g/cc which is close

to the theoretical value of 2.71 for pure aluminum. Thus, with this decomposition reaction, it is possible to synthesize aluminum particles in the form of cubes.

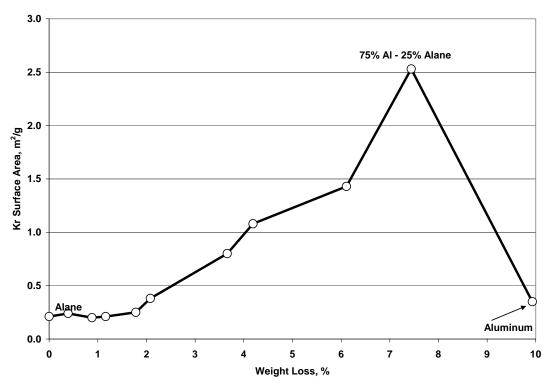


Fig. 4. Dependence of surface area of alane on extent of hydrogen evolution.

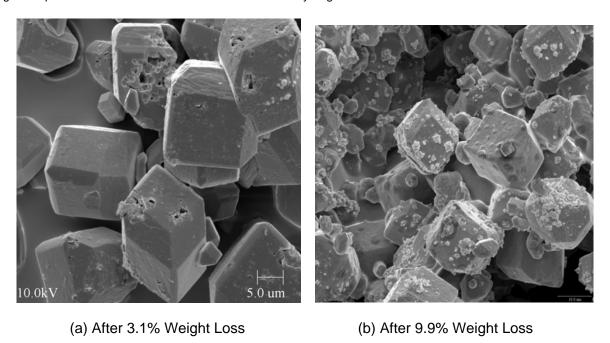


Fig. 5. SEM photos for (a) partially decomposed alane (3.1% weight loss) and (b) completely decomposed alane (~ 10% weight loss).

## 3.5 Kinetics of alane decomposition in argon

The TGA results for alane decomposition in Ar at six heating rates (0.49, 0.98, 2.0, 4.9, 9.9 and 20.2°C/min) are shown in Figure 6. At 20.2°C/min, the decomposition began at 180 C and was completed at 215°C, and the final weight loss was 9.9%; indicating that the entire hydrogen content of alane, bonded to aluminum, has been completely removed. For the three repeated runs performed at 9.9°C/min, the reproducibility of the data was quite satisfactory and the average weight loss was 9.05%. The decomposition began at a lower temperature (168°C) and was completed at 197°C. This trend continued with the other heating rates. In all cases, the starting and final decomposition temperatures were lowered, as expected, when the heating rate decreased. The main noticeable trend is that as the heating rate decreased, the final weight loss became smaller. That is, a sample with a low heating rate retained more hydrogen than another one decomposing at a higher heating rate. It appears that when alane decomposes at a lower heating rate, the porous structure closes earlier and a portion of the hydrogen stays trapped inside the particles. Also, at a lower heating rate, decomposition occurs at lower temperatures; the diffusion of hydrogen through the particles or nucleation of aluminum sites is slow.

To perform kinetic analysis in Ar atmosphere, we have pursued two independent approaches: the classical "model free kinetics" (MFK), and a more precise curve-fitting thermo-kinetics modeling (TKM) analysis. The classical MFK approach was used first to estimate the activation energy and the pre-exponential factor related to the decomposition reaction. For these classical methods, the assumption is usually made at some point of the analysis that the decomposition is a first order reaction. Based on the results of analysis with MFK, the more sophisticated TKM approach is then used with as many necessary intermediate kinetic steps and reaction orders as needed to mathematically describe the steps involved. The model showing the lowest possible standard deviation was assigned to the decomposition reaction. In the present investigation, the kinetic parameters were estimated for different types of reactions, and finally, physical models for the overall reaction were proposed.

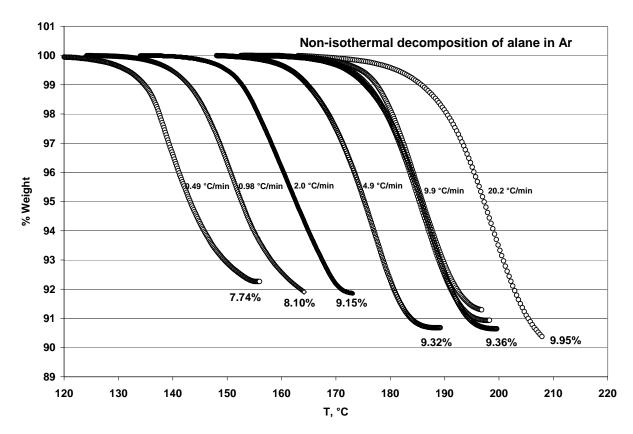


Fig. 6. Decomposition of alane in Ar (50 cc/min) at different heating rates.

3.5.1 Model free kinetics (MFK) - isoconversion method

The classical MFK computations include the three main well-established independent methods: the ASTM method [23], the Friedman analysis [24], and the Ozawa-Flynn-Wall analysis [25,26]. With the ASTM method, an Arrhenius plot is made between the logarithm of heating rates versus reciprocal of absolute temperatures at the points when the maximum decomposition rate occurs at each heating rate. If the plot is linear, the slope and intercept would yield the activation energy,  $E_a$ , and the pre-exponential term, A. Figure 7 displays the results obtained with this method; the calculated values for alane decomposition are:  $E_a = 97.0 \pm 3.1 \text{ kJ/mol}$  and  $A = 2.04 \times 10^{-8} \text{ s}^{-1}$ .

The second MFK method is the Friedman analysis [24] which is based on the iso-conversion method: at a given level of decomposition, the instantaneous decomposition rate and its corresponding temperature are calculated for each heating rate. This step is repeated at other decomposition levels until the reaction is completed. For each decomposition level (conversion), the plot between the logarithm of decomposition rate and the reciprocal of decomposition temperature would normally yield a straight line, the slope and intercept of which gives respectively the values of  $E_a$  and A at this particular level of decomposition. The computation is repeated for all other levels of decomposition and several lines, corresponding to several iso-conversion levels, are produced. The corresponding values of  $E_a$  and A, are obtained for each level of decomposition. The results are summarized in Figure 8. When the reaction involves more than one step, the contribution of each step to the overall kinetics will influence the observed activation energy. Thus, Figure 8 simply suggests that alane decomposition follows, at least, two steps: the first starts and continues up to  $\sim 0.25$  conversion ( $\sim 2.5\%$  weight loss for alane) and the second occurs between 0.25 and 0.85 conversion ( $\sim 2.5$  to 8.5% weight loss for alane). This finding is in line with the results of specific surface area reported above which are displayed in Figure 4.

The third MFK method is the Ozawa-Flynn-Wall analysis [25,26] which also utilizes the iso-conversion approach. Similar to the Friedman analysis, a given conversion is first chosen. For each heating rate, the corresponding temperature is noted. A plot between the logarithm of heating rate versus reciprocal of

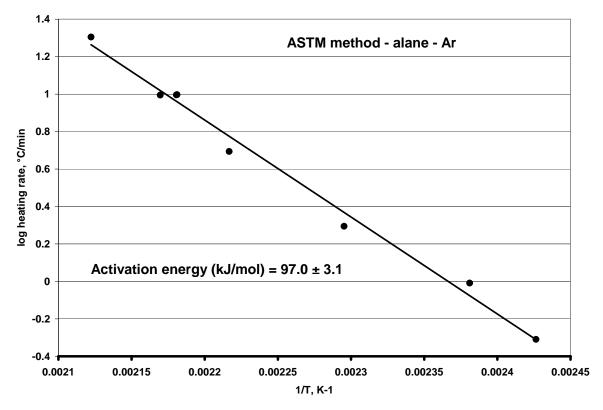


Fig. 7. Arrhenius plot for decomposition of alane in Ar using the ASTM method.

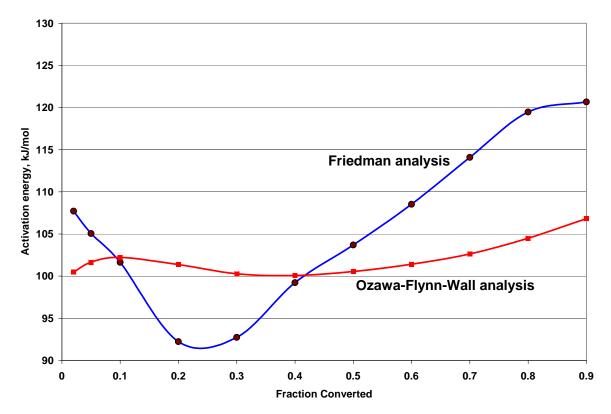


Fig. 8. Variation of activation energy with level of alane conversion to yield aluminum.

absolute temperature is made. A linear relationship is normally obtained and the corresponding values of  $E_a$  and A are computed for this particular conversion. The computation is repeated at other levels of conversion and a plot is finally made between  $E_a$  and conversion (Figure 8). This plot for alane decomposition can be interpreted in two different ways. First, one may argue that with this method, the activation energy of alane decomposition is constant through the entire reaction with an average value of  $103 \pm 3$  kJ/mol. In this case, the decomposition of alane would follow first order kinetics. This is not the case as it will be shown shortly. Second, as was the case with the Friedman analysis, one may also argue that alane decomposition is more complicated and involves more than one step. Both scenarios will be considered next.

#### 3.5.2 Thermokinetics Modeling

Additional attempts were made to fit the original raw data to different kinetic models. The simplest model involves first order decomposition kinetics; that is: Alane  $\rightarrow$  Aluminum + H<sub>2</sub>. The rate of this reaction would be given by:  $(d\alpha/dt) = A * (1-\alpha) * exp (-E_a/RT)$ , where  $\alpha$  is the conversion fraction and R is the gas constant. The analysis indicated that alane decomposition does not follow first order kinetics as shown by the dotted (straight) lines in Figure 9.

In the next attempts, more complicated two-step models were considered; the most promising included two-step consecutive reactions of the type:  $A \rightarrow B \rightarrow C$ . The two best fitting models have been chosen, they are referred to here as Model-A and Model-B. Both models were statistically near perfection with correlation coefficients better than 0.998. In both cases, step-1 ( $A \rightarrow B$ ) was basically the same; it was governed by the Kolmogorov-Johnson-Mehl-Avrami (KJMA) equation (a modified Avrami equation), which describes mainly an  $n^{th}$ -dimensional nucleation or growth reaction [20, 27-29]. The KJMA model has been represented recently by the following equation proposed by Nakamura [29]:

$$d\alpha/dt = n * k(T) * (1 - \alpha) * [-Ln(1 - \alpha)]^{(n-1)/n}$$

where k(T) is the rate constant, given by: k(T) = A \* Exp(-E/RT). Here A is the pre-exponential factor and E is its activation energy. The Avrami exponent, n, is normally used as a tracer for the dimensionality of the reaction. Commonly accepted guidelines state that for one-dimensional growth,  $n \le 2$ ; for two-dimensional growth, n = 2 - 3 and for 3 dimensional growth, n = 3 - 4 [29, 30]. Values of the kinetic parameters obtained here for step-1 in Model-A and Model-B are listed in Table 2.

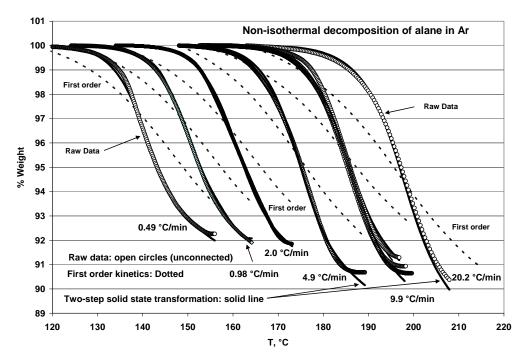


Fig. 9. Comparison between the experimental data of alane decomposition (open circles), first order decomposition kinetics model (dotted lines) and two-step reaction model (solid-line).

Table 2: Calculated kinetic parameters for non-isothermal decomposition of alane in argon.

	Model-A	Model B
Step-1		
<i>Step-1</i> A <sub>1</sub> , s <sup>-1</sup>	5.61 * 10 <sup>7</sup>	9.18 * 10 <sup>8</sup>
E <sub>1</sub> , kJ/mol	83.8	99.0
Avrami exponent: n₁	2.23	2.43
Step-2		
<i>Step-2</i> A <sub>2</sub> , s <sup>-1</sup>	4.92 * 10 <sup>5</sup>	3.08 * 10 <sup>9</sup>
E <sub>2</sub> , kJ/mol	63.1	88.5
Avrami exponent: n <sub>2</sub>	0.63	N/A
reaction order	N/A	2.52

The second steps for Model-A and Model-B can be equally described statistically by two different equations. For Model A, step-2, satisfied, once again, a second KJMA equation; the parameters of which are displayed in Table 2; they are lower than those of step-1. However for Model-B, the kinetics satisfied a typical chemical reaction (with  $n^{th}$  order kinetics) of the type:  $d\alpha/dt = A_2 * (1-\alpha)^n * Exp(-E_2/RT)$ . In this case, the order of the reaction (power of the conversion term) was 2.52. The two models gave excellent fitting to the raw data as indicated by the solid lines shown in Figure 9 (Because the fitting is excellent, it may be hard to see the complete solid lines at certain regions of the plots since the solid lines are perfectly superimposing the raw data points). From the statistical analysis, correlation coefficients and

standard deviation values, Model-A appeared slightly better than Model-B. However, Model-B could not be ruled out because its fitting to the data was also excellent.

Based on Model-A, step-1 has a higher activation energy, pre-exponential term and Avrami exponent than Step-2 (Table 2). From the values of A, E, and n, obtained with each step, the KJMA equation was used to calculate the rate of each individual reaction,  $d\alpha/dt$ , at different temperatures and several values of  $\alpha$ . Figure 10 shows the dependence of step-1 rates on conversion at 5 chosen temperatures. The rates increase with conversion, reach a maximum at  $\alpha = 0.45$  and finally decrease when  $\alpha > 0.45$ . This trend is typical of a nucleation process. The Avrami exponent,  $n_1$  is between 2 and 3 which suggests that the first step for alane decomposition is a two-dimensional nucleation reaction possibly taking place at the outer surface of the particles (two dimensional nucleation).

For Model-A, step 2 has lower values of A, E and n than step-1. The rates for this step are much faster than in step-1. Figure 11 shows that the rates of step-2 increase with temperature but decrease with increasing coverage. This step is attributed to a growth reaction; the growth of aluminum layers on the nucleation sites already developed in step-1. The small value of  $n_2$ , which is < 1, suggests that the growth reaction is a one-dimensional process, most likely, starting at the outer surface of the particles and continuing inwards towards their center.

A comparison between the rates of nucleation and growth at 100°C is illustrated in Figure 12. This trend was valid at all other temperatures. Since the growth rate is considerably higher than the nucleation rate, we conclude that the nucleation (surface) reaction is the rate determining step that controls the overall decomposition of alane. This suggests that if the nucleation sites are or chemically stabilized on the outer surface of the particles, the nucleation rate will be reduced and the rate of alane decomposition will be slowed.

As mentioned earlier, with Model-B, step-1 was similar to that of Model-A, however, step-2 was not the same; it followed a simple rate law expression of the type:  $d\alpha/dt \sim (1 - \alpha)^n$ , where n (value = 2.52) is the overall order of the second step. One possible explanation is that the aluminum atoms, once formed after

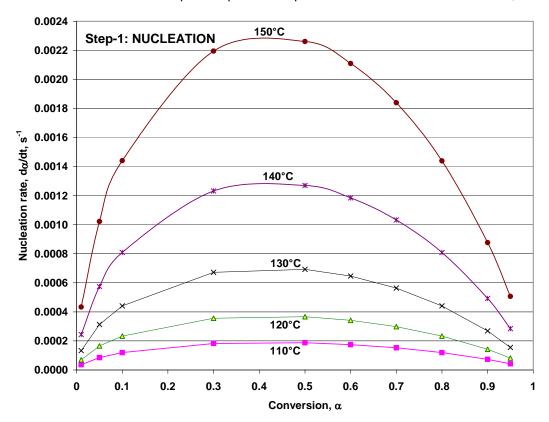


Fig. 10. Dependence of nucleation rates on conversion and temperature.

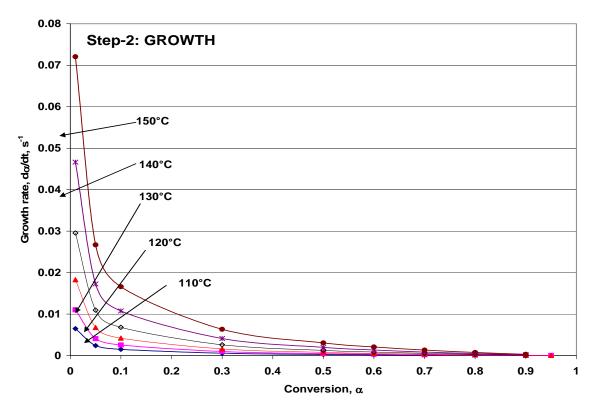


Fig. 11. Dependence of growth rates on conversion and temperature.

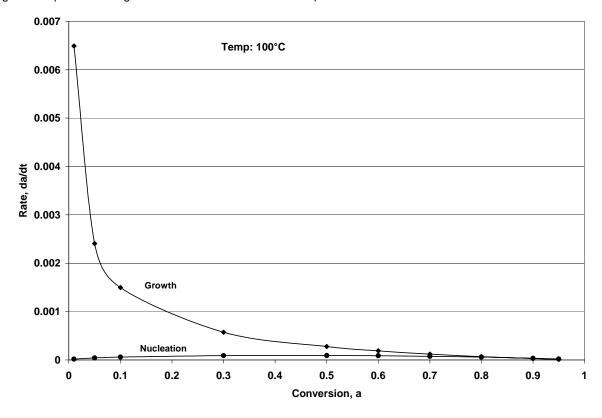


Fig. 12. Comparison between nucleation and growth rates at 100  $^{\circ}\text{C}.$ 

step-1, may move over the surface and collide with each other to form metallic bonds or to enter the crystalline structure (FCC for aluminum). With this assumption, the aluminum atoms have to collide with each other at a rate of 2 -3 times, on the average, before they bond to each other. If this is the case, the number of collisions at the beginning of growth process would be high (say 3 or 4) and it will decrease, due to the proximity of aluminum atoms, as the growth process continues. Further qualification for this mechanism is beyond the scope of this article and is left to future research.

#### 3.5 Model Prediction

With the two models developed, the prediction of alane shelf-life during storage at 5, 10, 15, 20, 25 or 30°C was computed. The results, which are displayed in Table 3, show two significant trends. First, Model-A predicts longer shelf-life than Model-B. For example, after 3 years storage at 20°C, while Model-A predicts 0.13% weight loss of hydrogen, however, Model-B predicts 1.02% weight loss. It is recalled from Table-1 that this sample had 9.95% elemental hydrogen content which is smaller than the theoretical value of 10.08% by 0.13%. Since this sample has been stored in our laboratories at ambient temperature for ~ 3 years, one concludes that Model-A appears more realistic for predictions than Model-B. Second, the prediction shows clearly how sensitive alane decomposition is to storage temperature fluctuations. For example, if alane is stored for 10 years at 5-10°C, the amount decomposed (as per Model-A) is insignificant; at 15°C the sample loses 0.31% and at 30°C it loses most of its hydrogen (8.61%). Thus, the important critical element to extend the shelf life of alane is to store it at low temperatures, preferably in the neighborhood of 10°C.

Table 3: Prediction of weight percent of hydrogen evolved during storage of alane

(a)	\ R /	-	$\sim$	١ ٨
la.	JΙV	IUU		

Years	5°C	10°C	15°C	20°C	25°C	30°C
1	0.00	0.00	0.01	0.03	0.08	0.28
3	0.00	0.01	0.03	0.13	0.76	3.39
5	0.00	0.01	0.07	0.42	2.43	6.37
10	0.01	0.04	0.31	2.10	6.27	8.61
15	0.02	0.10	0.85	4.29	7.86	9.18

## (b) Model-B

Years	5°C	10°C	15°C	20°C	25°C	30°C
1	0.00	0.01	0.03	0.10	0.42	1.74
3	0.01	0.05	0.22	1.02	3.52	5.79
5	0.03	0.14	0.68	2.76	5.27	7.22
10	0.12	0.62	2.67	5.16	7.16	9.02
15	0.29	1.47	4.19	6.24	8.28	9.77

#### 4. Conclusions and Recommendations

- 1. The decomposition of alane starts with hydrogen liberation from external surface and pre-existing pores and cracks. Porosity then increases while the size of alane particles decreases. When the decomposition is completed, the pores are closed in the final product (aluminum).
- 2. The analysis of non-isothermal decomposition of alane in Ar was performed using a popular ASTM kinetic model. This approach is useful perhaps in yielding a reasonable single global value of activation energy (and pre-exponential term) that can be compared to other literature values.
- 3. With Freidman MFK analysis, however, alane decomposition involves two steps; this finding is supported by the results of surface area measurements.
- 4. Alane decomposition can be better described by solid-state transformation kinetics (nucleation and growth mechanism). Two kinetic models have been developed and their fit to the experimental data was confirmed. The standard deviations for the fitting were better than 0.998.

- The models were used to estimate the shelf-life of alane when stored in inert atmospheres for several years.
- 5. Alane decomposition in Ar involves two major consecutive steps: A → B → C; representing the transformation of hexagonal alane crystals to amorphous aluminum particles (X-ray diffraction of the aluminum left after decomposition did not yield any peaks). Step-1 is due to a slow nucleation reaction; possibly with the formation of aluminum nucleation sites at the outer surface of the particles. This is the rate determining step for alane decomposition and the kinetics are controlled by an Avrami-type (KJMA) equation. The second step is postulated as the growth of aluminum layer towards the center of the particles (step-2).
- 6. Based on the mechanism of alane decomposition proposed here, we may suggest possible methods to further slow down the decomposition of alane. This includes coating the particles with inert materials or formation of a protective oxide layer at the outer surface.
- 7. With Model-A developed, the projected decomposition of alane in 15 years (or less) is very small, ~ 0.1%, if stored below 10°C. Storing alane at higher temperatures is not recommended; the material may lose as much as its entire hydrogen content if stored for 15 years at 30°C.

#### Acknowledgement

The authors want to thank Dr. Ronald Channell, Dr. Frank Roberto, Dr. Robert Corley, Dr. Keith McFall, Mr. John Clark, Mr. Paul Jones, Dr. George Harting and Dr. Donald Tzeng for their fruitful discussions, suggestions and support. The technical assistance of Ms Marietta Fernandez (SEM examinations) and of Ms Leslie Hudgens (density measurements) is highly appreciated.

I. M. K. Ismail is appreciative to late Professor J. Opfermann for his invaluable advice and the effort he made in modifying the kinetics software to meet our needs.

#### References

- 1. F. M. Brower, N. E. Matzek, P. F. Reigler, H. W. Rinn, C. B. Roberts, D. L. Schmidt, J. A. Snovar, K. Terada; *J. Am. Chem. Soc.* 98(9) (1976) 2450-2453.
- 2. M. Taliani; Gazz. Chim. Ital. 51 (1921) 184.
- 3. G. C. Sinke, L. C. Walker, F. L. Oetting, D. R. Stull; J. Chem. Phys.47 (1967) 2759.
- 4. M. A. Petrie, J. C. Bottaro, R.J. Schmitt, P. E. Penwell, D. C. Bomberger; *United States Patent* US 6228338 B1 (2001).
- 5. V. V. Zakharov, A. N. Tskhai; Russian Journal of Inorganic Chemistry 37 (9) (1992) 997.
- 6. V. P. Tarasov, Yu. B. Muravlve, S. I. Bakum, A. V. Novikov; Doklady Phys. Chem. 393 (2003) 4.
- 7. S. V. Vvazovkin, A. I. Lesnikovich: Thermochimica Acta 165 (1990) 273-280.
- 8. S. Vyazovkin, V. Goryachko; Thermochimica Acta 194 (1992) 221-230.
- 9. S. Vyazovkin, C. A. Wight; Ann. Rev. Phys. Chem. 48 (1997) 125-149.
- 10. J. Opfermann, W.Haedrich: Thermochimica Acta 263 (1995) 29 50.
- 11. S. Vyazovkin, C. A. Wight; *Thermochimica Acta* (1999) 340-341 53-68.
- 12. S. Vyazovkin; Int. Rev. Phys. Chem. 19 (2000) 45-60.
- 13. S. Vyazovkin, C. A. Wight; Chem. Mater 11 (1999) 3386-3393.
- 14. S. Vyazovkin, J. S. Clawson, C. A. Wight; Chem. Mater. 13 (2001) 960-966.
- 15. S. Vyazovkin, C. A. Wight; J. Phys. Chem. A 101 (1997) 8279-8284.
- 16. S. Vyazovkin, C. A. Wight; J. Phys. Chem. A 101 (1997) 5653-5658.
- 17. H. J. Flammersheim, J. Opfermann.; Thermochimica Acta 337 (1999) 149-153.
- 18. J. Opfermann, W.Haedrich: Thermochimica Acta 263 (1995) 29 50.
- 19. B. Roduit, C. Borgeat, B. Berger, P. Folly, B. Alonso, J.N. Aebischer; *Proceedings of the 31<sup>st</sup> International Pyrotechnics Seminars (*2004) 11-16July,Fort Collins, Colorado, USA.
- M. Avrami; J. Chem. Phys. 7 (1993) 1103-1112, 8 (1939) 212-224, 9 (1940) 177-184.
- 21. I. M. K. Ismail; Langmuir 8 (1992), 360.
- 22. Turley, J. W and Rinn, H. W; Inorg. Chem. 8 (1969) 18.
- 23. H. E. Kissinger; J. Res. Nat. Bur. Stds 57 (1956) 217.
- 24. H. L. Friedman; J. Polymer Lett. 4, (1966) 323.
- 25. T. Ozawa; Bull. Chem. Soc., Japan 38, (1965) 1881.
- 26. J. Flynn, L. A. Wall; Polymer Lett. 4, (1966) 232.

- 27. A. N. Kolmogorov; Izvestiya Acad. Nauk SSSR, Ser. Math. 1 (1937) 355.

- 28. W. A. Johnson, P. A. Mehl; *Trans. Am. Inst. Mining Metall. Eng* 135 (1939) 416.
  29. N. Nakamura, K. Katayama, T. Amano; J. Appl. Polym. Sci 17 (1975) 1031.
  30. T. Pradell, D. Crespo, N. Clavaguera, M. T. Clavaguera-Mora; *J. Phys. : Condens. Matter* 10 (1998) 3833.